

FIG. 1. Total emissivity of ammonia.

as calculated from this correlation and extrapolation procedure is shown in Fig. 1 in comparison with the early *. findings of Port. It should be noted that in Fig. 1 p_a stands for the ammonia partial pressure, and L for the geometric mean beam length. The agreement between the suggested values of Port and the present prediction is quite good indeed. It is a little surprising, however, to see that the agreement in the extrapolated (higher temperatures) region

is better than that at 300° K, since the prediction at 300° K should be most reliable as it is purely based on the France-Williams band absorption data.

REFERENCES

- 1, F. J. PORT, Heat transmission by radiation from gases, Sc.D. thesis in Chemical Engineering, Massachusetts Institute of Technology, Cambridge, Massachusetts (1940).
- 2. H. C. **HOITEL** and A. F. SAROFIM, *Radiutiue Transfer,* p. *236.* McGraw-Hill, New York (1967).
- 3. G. HERZBERG, *Infrared and Raman Spectra of Polyatomic* Molecules. Van Nostrand, New York (1945).
- 4. D. C. MCKEAN and P. N. SCHATZ, Absolute infrared intensities of vibration bands in ammonia and phosphine, *J. Chem. Phys.* 24, 316 (1956).
- W. L. FRANCE and D. WILLIAMS, Total absorptance of ammonia in the infrared, *J. Opt. Sot. Am. 56, 70 (1966).*
- T. E. WALSH, Infrared absorptance of ammonia-20 to 35 microns, *J. Opt. Sot. Am. 59,261 (1969).*
- *C.* L. TIEN, Thermal radiation properties of gases, Advances in Heat Transfer, edited by J. P. HARTNETT and T. F. IRVINE, Vol. 5, pp. *253-254.* Academic Press, New York (1968).
- D. K. EDWARDS and A. BALAKRISHNAN, Thermal radiation by combustion gases. Int. *J. Heat Muss Transfer* **16, 2540** (1973).
- 9. S. H. CHAN and C. L. TIEN, Infrared radiation properties of sulfur dioxide, *J. Heat Transfer 93,* 172 (1971).
- 10. C. L. TIEN, M. F. MODEST and C. R. MCCREIGHT, Infrared radiation properties of nitrous oxide, J. Quant. *Spectosc. Radial. Transfer 12, 267 (1972).*

Int. J. Heat Mass Transfer. Vol. 16, pp. 857-861. Pergamon Press 1973. Printed in Great Britain

HEAT TRANSFER PARAMETERS OF A PARALLEL PLATE HEAT EXCHANGER

V. M. K. SASTRI **and** K. MASTANAIAH*

Indian Institute of Technology, Madras, India

(Received 10 *May* 1972 *and in revisedform 8 September* 1972)

a, distance between the plates through which laminar flow occurs;

b,

 C_i specific heat of fluid, *i*;

NOMENCLATURE 9. dimensionless velocity distribution of the laminar
between the plates through which laminar
 $\qquad -\text{side fluid. } u/\bar{u}$:

wall thickness;
 $*$ Presently Engineer, Space Science and Technology,
 $*$ Presently Engineer, Space Science and Technology,
 $*$ Privandrum, India.

- Graetz function ; G,
- h_{i} heat transfer coefficient of fluid i ;
- Н. heat capacity flow rate ratio, $C_2 W_2 / C_1 W_1$;
- thermal conductivity of the laminar-side fluid; k_1 , K. relative thermal resistance of the common wall and turbulent side convection ;
- $k_\mathrm{w},$ wall thermal conductivity;
- $l,$ axial length measured from the laminar side inlet ; L. overall length of the exchanger:
- laminar side Nusselt number, 2a h_1/k ; Nu_{1}
- overall Nusselt number, 2a *U,/k;* Nu_1°
- normalised laminar-side Nusselt number; N_1 ,
- laminar-side Péclét number, $2a \bar{u}/\alpha$; Pe,
- temperature of fluid i; t_i
- laminar side inlet temperature; $t_{1,0}$
- $t_{2,0}$ turbulent side inlet temperature;
- local axial laminar side fluid velocity; и,
- average laminar side fluid velocity; ū,
- overall heat transfer coefficient referred to laminar U_1 side ;
- W_{i} mass rate of flow of fluid i ;
- dimensionless transverse position, v/a ; \mathbf{x}
- transverse position measured from the bottom у, plate;
- dimensionless axial position *(2/Pe)(l/a);* z,
- dimensionless heat exchanger length *(2/Pe)(L/a)* ; Z.
- thermal diffusivity of laminar side fluid ; α .
- heat exchanger effectiveness; ε ,
- dimensionless temperature of fluid *i,* ξ_{i}

 $(t_i - t_{2,0})/(t_{i,0} - t_{2,0});$

- Λ . additional heat exchanger length ;
- \ldots (∞), fully developed values.

Subscripts

- 1, laminar side;
2, turbulent side
- turbulent side;
- ∞ , fully developed value.

THE **DESIGN** of heat exchangers, is customarily based on the tions. This assumption is reasonably valid only for turbulent flow of fluids [1, 2]. But for fluids in laminar flow the heatassumption of uniform heat-transfer coefficient along the transfer coefficients become sensitive to the actual boundary length of the exchanger, irrespective of the boundary condiconditions and may not be sufficiently uniform along the length of the exchanger. Also thermal entrance regions for laminar flow can be significant $\lceil 1, 3-5 \rceil$.

Stein and Sastri [6] recently presented a detailed analysis of heat exchanger with laminar tube-side and turbulent shell-side flows as a new extension of the classical Graetz problem and reported various quantities relating to cocurrent and countercurrent flows. They assumed uniform heat transfer coefficient on the shell-side and solved the twodimensional energy equation for the tube-side fluid and showed that predictions can be made by use of the actual fully developed heat transfer coefficient and an effective heat exchanger length and that both of these quantities depend on the operating conditions of the exchanger.

The present note applies the above analysis to a parallel plate heat exchanger with laminar flow on one side and turbulent flow on the other. The fully developed Nusselt number and the thermal entrance length are given as functions of operating parameters for both cocurrent and countercurrent flows.

ANALYSIS

A schematic of the parallel plate exchanger is shown in Fig. 1. Assuming a constant heat-transfer coefficient on the turbulent side, the appropriate laminar side energy equation is written in dimensionless form as

$$
\frac{\partial^2 \xi_1}{\partial x^2} = g(x) \frac{\partial \xi_1}{\partial z}, \qquad \xi_1(x, z) : 0 \le x \le 1 \tag{1}
$$
\n
$$
0 \le z \le Z
$$

FiG. 1. Schematic diagram of a parallel plate heat exchange

where

$$
g(x) = 6x(1 - x). \tag{2}
$$

Here the equivalent Graetz function $G(\lambda, x)$ is assumed to be of the form

$$
G(\lambda, x) = \sum_{n=1}^{\infty} A_n(\lambda) x^n
$$
 (3)

with the recurrence relation given by

$$
A_0 = 1, A_1 = 0, A_2 = 0
$$

and

$$
A_n = \frac{6\lambda}{n(n-1)} \left[A_{n-4} - A_{n-3} \right] \text{ for } n \geq 3. \tag{4}
$$

NUSSELT NUMBERS

It can be shown that for sufficiently large z, the laminar side Nusselt number can be given by

$$
Nu_1(\infty) = 2H\lambda_1/(H + \delta - KH\lambda_1) \tag{5}
$$

where λ_1 is the first order eigen value of the characteristic equation and the corresponding overall Nusselt number is given by

$$
Nu_1^{\circ}(\infty) = 2H\lambda_1/(H+\delta). \tag{6}
$$

For the special case of $\delta = -1$ and $H = 1$, it is known that

$$
Nu_1(\infty) = 70/13 = 5.385\tag{7}
$$

which corresponds to the boundary condition of uniform wall heat flux. The corresponding $Nu_1^{\circ}(\infty)$ is given by

$$
Nu_1^{\circ}(\infty) = 70/(13 + 35K). \tag{8}
$$

It may be noted that the uniform wall temperature boundary condition is attained as $H \to \infty$ and $K \to 0$ (or $KH \rightarrow 1$). For this case λ_1 is found to be 2.4303 and the corresponding

$$
Nu_{1}(\infty) = Nu_{1}^{c}(\infty) = 2\lambda_{1} = 4.8606.
$$

ADDITIONAL **HEAT EXCHANGER LENGTH**

The traditional definition of NTU may be modified by writing

$$
NTU = \frac{1}{2} Nu_1^{\circ}(\infty) [Z + \Delta]
$$
 (9)

for this geometry, where Δ is the appropriate additional heat exchanger length, which takes into account the effects of thermal entrance regions. Thus, we obtain the relations for *A* as follows: (except for $\delta = -1$, $H = 1$)

$$
\Delta = -\frac{1}{\lambda_1} \ln \left[-\frac{H+\delta}{H} \sum_{n=1}^{\infty} B_n \exp \left\{ -(\lambda_n - \lambda_1) Z \right\} \right] (10)
$$

and

$$
\Delta_{\infty} = -\frac{1}{\lambda_1} \ln \bigg[-\bigg(\frac{H+\delta}{H}\bigg) B_1 \bigg]. \tag{11}
$$

For the case of uniform wall heat flux $(\delta = -1, H = 1)$,

$$
\Delta = \Delta_{\infty} + \left(\frac{26 + 70 \text{ K}}{70}\right) \sum_{n=1} B_n \exp\left(-\lambda_n Z\right) \tag{12}
$$

with

$$
\Delta_{\infty} = \frac{4454}{3 \times 4 \times 11 \times 35 \times (26 + 70 \text{ K})}.
$$
 (13)

A is, in general, a function of *H, K* and the mode of operation and the length of the exchanger Z. However, in most cases of practical interest where the effectiveness is greater than 0.5, Z is sufficiently large such that Δ_{∞} would be sufficient for most applications [3].

RESULTS AND DISCUSSION

The factors that are directly related to the overall heat transfer rates are the laminar side fully developed Nusselt number $Nu_1(\infty)$ and the additional heat exchanger length $\varDelta_{\infty}.$

In Fig. 2, $Nu_1(\infty)$ is normalised with respect to the value corresponding to the case of uniform wall heat flux and shown as a function of *H* and *K.* The normalized value corresponding to isothermal wall is about 0.83 and is shown in Fig. 2. The behaviour is qualitatively identical to that found with other heat exchanger analyses [3, 4, 6]. In general, it is observed that operating conditions have significant effect on laminar side heat transfer coefficients $Nu_1(\infty)$ is smaller for cocurrent flow than for counter-current

FIG. 2. Variation of normalized Nusselt number with operating parameters.

flow. In cocurrent flow, the fully developed coefficients are never larger than the value corresponding to the uniform wall heat flux boundary condition. On the other hand, in countercurrent flow, fully developed coefficients are never smaller than the value corresponding to the uniform wall temperature boundary condition, but can be significantly larger than the value corresponding to the boundary condition of uniform heat flux.

Figure 3 shows the dependence of A_{∞} on the operating parameters. It is seen that the values of A_{∞} for cocurrent flow are greater than for countercurrent flow, indicating that thermal entrance region is more significant with cocurrent flow than with counter flow.

For values of *K* less than about 1.0 and for given *H*, Fig. 2. shows that the order of magnitude of error in $Nu_1(\infty)$ of the traditional design formulae is the same for both flow arrangements. However, the error can be greatly magnified for countercurrent flow because A_{∞} is much smaller as can

FIG. 3. Variation of additional heat exchanger length with

be seen in Fig. 3. This is also evident from equation (9) in which NTU is given as a product of $Nu_1(\infty)$ and *A*. This influence is even more significant for decreasing values of H .

Tables 1 and 2 give the computed quantities for cocurrent and countercurrent flows respectively. In each table are shown the laminar side fully developed Nusselt number, the additional heat exchanger length A_{∞} , Z^* and ε^* . Z^* is the value of Z at which $A = A_{\infty} (A = 0.95 A_{\infty})$ as defined in [6] and ε^* is the corresponding effectiveness. Heat exchanger computations with $A = A_{\infty}$ require that $Z \ge Z^*$ or $\varepsilon \ge \varepsilon^*$.

These latter conditions *are* satisfied **as can be seen in the** Tables 1 and 2 so that the use of A_{∞} is justified for most practical applications.

1. C. A. SLEICHER and M. TRIBUS, Heat Transfer in a pipe operating parameters. distribution, *Trans. Am. Soc. Mech. Engrs* **79**, 789-797.

Table 1. Quantities *related to the cocurrent flow* Table 2. Quantities *related to the countercurrent flow*

H	$Nu_1^{\circ}(\infty)$ $Nu_1(\infty)$		\mathcal{A}_{∞}	Z^*	ε^*	Η		$Nu_1^{\circ}(\infty)$ $Nu_1(\infty)$	Δ_{∞}	Z^*	ε^*	
		$K = 0.1$					$K = 0.1$					
0 ₁	1.912	2.113	0.1143	0.109	0.895	0.1	6.064	8.703	0.0072	0.039	0.737	
0.5	3.440	4 1 5 5	0.0502	0.138	0.616	0.5	4.500	5.807	0.0242	0.105	0.399	
1 ₀	3.706	4.549	0.0423	0.134	0.474	$1-0$	4.242	5-385	0.0292	0.116	0.233	
2.0	3.842	4754	0.0387	0130	0.381	2 ₀	4.110	$5-173$	0.0321	0.121	0.252	
10 ₀	3.948	4.920	0.0359	0.127	0.294	$10-0$	4.022	5:004	0.0346	0.125	0.267	
					$K = 0.5$							
0 ₁	1.646	$2 - 797$	0-0658	0.201	0.906	$0-1$	2.566	7.152	0.0063	0.053	0.516	
0.5	1.154	4.669	0.0235	0.155	0.435	0.5	2.336	5.610	00140	0.111	0.237	
$1-0$	2.204	4.913	0.0205	0.144	0.301	1 ₀	2.295	5.385	0.0158	0.122	0.135	
20	2.228	$5-033$	0.0192	0.138	0.229	2.0	2.274	5.269	0.0168	0.127	0.144	
$10-0$	2.248	5.128	0.0182	0.134	0.169	100	$2 - 256$	5.176	0.0177	0.131	0.152	
			$K = 1.0$									
$0-1$	1.282	3.563	0-0309	0.213	0.816	0 ₁	1.532	6.542	0.0052	0.045	0.307	
0.5	1.422	4.929	00132	0.124	0251	0.5	1.468	5.528	0.0092	0.089	0-128	
1·0	1.436	5.085	0.0120	0.115	0.164	$1-0$	1.458	5.385	0:01	0.097	0.071	
20	1.442	5.162	0.0115	0.110	0.121	$2-0$	1.452	5.311	0.0105	0.101	0-075	
$10-0$	1.446	5.122	0.0111	0.106	0.088	$10-0$	1.448	5.252	0.0109	0.105	0.079	
	$K = 2.0$						$K = 20$					
0:1	0.812	4.347	0.1140	0.114	0.420	$0-1$	0.858	6.080	0.0038	0.017	0.074	
0.5	0.836	5.124	0.0068	0.059	0.076	0.5	0.846	5.468	0.0055	0.040	0-036	
$1-0$	0.840	5.213	0.0065	0.053	0.047	$1-0$	0.843	5.385	0.0058	0.044	0.019	
$2 - 0$	0.840	5.256	0.0063	0.051	0.034	$2 - 0$	0.842	5.342	0.0060	0.047	$0 - 021$	
$10-0$	0.842	5.291	0.0062	0.049	0.024	$10-0$	0.842	5.308	0.0061	0.048	0.022	
$K = 10-0$							$K = 10.0$					
$0-1$	0.192	5.166	0.0015	$0-005$	0.005	$0-1$	0.194	5.554	0.0012	0-005	0.005	
0.5	0.192	5:326	00014	0.005	$0 - 001$	0.5	0.192	5.404	0.0013	0.005	0:001	
$1-0$	0.192	5.346	0.0014	0.005	0:001	$1-0$	0.193	5.385	0.0013	0.005	0:001	
$2 - 0$	0.192	5.356	0.0014	0.005	$0 - 001$	$2-0$	0.192	5.375	0.0013	0:005	0:000	
100	0.192	5.363	0.0013	0.005	0.001	10.00	0.192	5.367	00013	0.005	0.000	

- 2. R. SIEGEL and E. M. **SPARROW,** Comparison of turbulent heat transfer results for uniform heat flux and uniform wall temperature, J. *Heat Transfer* 82C, 152 (1960).
- 3. R. P. STEIN, Liquid metal heat transfer, *Advances in Heat Transfer,* edited by T. F. IRVINE and J. P. **HARTNETT,** Vol. III. Academic Press, New York (1966).
- 4. R. P. STEIN, Mathematical and practical aspects of heat transfer in double pipe heat exchangers, *Proceedings oj the Third International Heat Transfer Conference.* Vol. I, p. 139. A.I.Ch.E., New York (1966).
- 5. R. P. **STEIN.** The mathematics of counterflow heat exchangers with equal heat capacity flow rates, paper presented at the 9th National heat transfer conference, A.S.M.E.-A.I.Ch.E., Seattle, Washington (1956).
- 6. R. P. STEIN and V. M. K. SASTRI, A heat transfer analysis of heat exchangers with laminar tube side and turbulent shell side flows. A new heat exchanger Graetz problem. Presented at the Twelfth National Heat Transfer Conference A.I.Ch.E.-A.S.M.E. Tulsa, Oklahoma, August 15-18 (1971).